

Linear

Imperfections

CAS Zeuthen September 2003

Oliver Bruning / CERN AP-ABP

Linear Imperfections

equation of motion in an accelerator

→ Hills equation

→ sine and cosine like solutions + one turn map

dipole perturbations

→ closed orbit response

→ dispersion orbit

→ integer resonances

quadrupole perturbations

→ tune error

→ beta-beat

→ half-integer resonances

orbit correction

→ local orbit bumps

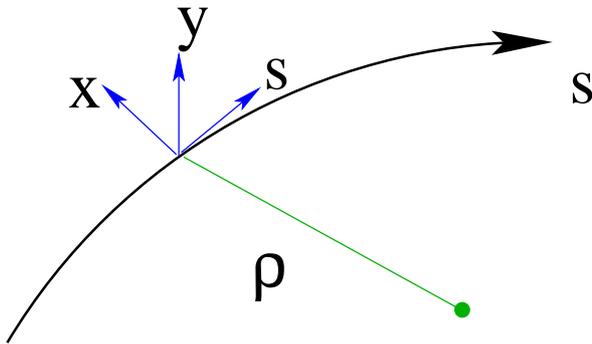
→ harmonic filtering

→ SVD

→ most effective corrector

Variable Definition

Variables in moving coordinate system:



$$\mathbf{x}' = \frac{d}{ds} \mathbf{x}$$

$$\frac{d}{dt} = \frac{ds}{dt} \cdot \frac{d}{ds} \rightarrow \mathbf{x}' = \frac{p_x}{p_0}$$

\swarrow
 \mathbf{v}

Hill's Equation:

$$\frac{d^2 \mathbf{x}}{ds^2} + \mathbf{K}(s) \cdot \mathbf{x} = \mathbf{0}; \quad \mathbf{K}(s) = \mathbf{K}(s + L);$$

$$\mathbf{K}(s) = \begin{cases} 0 & \text{drift} \\ 1/\rho^2 & \text{dipole} \\ 0.3 \cdot \frac{B[\text{T/m}]}{p[\text{GeV}]} & \text{quadrupole} \end{cases}$$

Sineline and Cosineline Solutions

■ ***system of first order***

linear differential equations:

$$\underline{y} = \begin{pmatrix} x \\ x' \end{pmatrix} \longrightarrow \underline{y}' + \begin{pmatrix} 0 & 1 \\ K & 0 \end{pmatrix} \cdot \underline{y} = 0$$

■ ***Floquet theorem:***

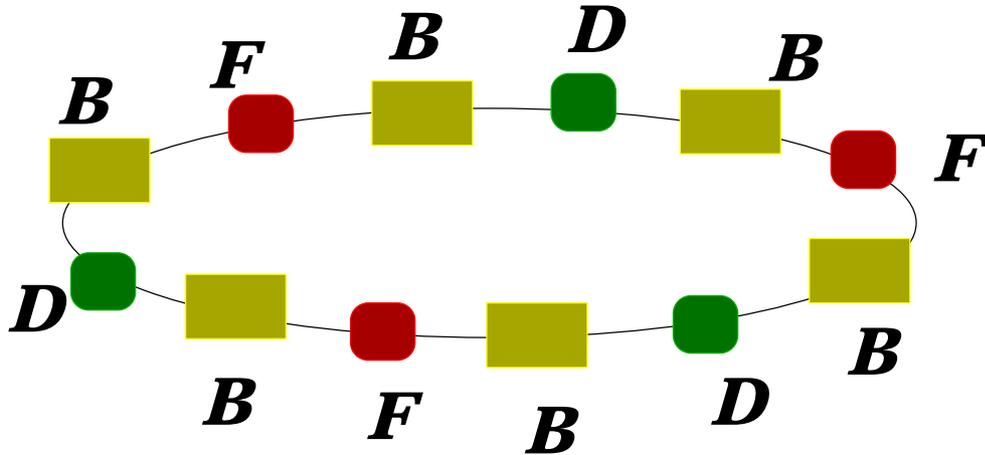
$$\underline{S}(s) = \begin{pmatrix} \sqrt{\beta(s)} \cdot \sin(\phi(s) + \phi_0) \\ [\cos(\phi(s) + \phi_0) + \alpha(s) \cdot \sin(\phi(s) + \phi_0)] / \sqrt{\beta(s)} \end{pmatrix}$$

$$\underline{C}(s) = \begin{pmatrix} \sqrt{\beta(s)} \cdot \cos(\phi(s) + \phi_0) \\ -[\sin(\phi(s) + \phi_0) + \alpha(s) \cdot \cos(\phi(s) + \phi_0)] / \sqrt{\beta(s)} \end{pmatrix}$$

$$\left(\beta(s) = \beta(s + L); \quad \phi(s) = \int \frac{1}{\beta} ds; \quad \alpha(s) = -\frac{1}{2} \beta'(s) \right)$$

Closed Orbit

- particles oscillate around an ideal orbit:



- additional dipole fields perturb the orbit:

■ error in dipole field

■ energy error

$$\alpha = \frac{l}{\rho} = \frac{\mathbf{q} \cdot \mathbf{B} \cdot \mathbf{l}}{\mathbf{p} + \Delta \mathbf{p}} \approx \left(1 - \frac{\Delta \mathbf{p}}{\mathbf{p}} \right) \cdot \frac{\mathbf{q} \cdot \mathbf{B} \cdot \mathbf{l}}{\mathbf{p}}$$

■ offset in quadrupole field

$$B_x = -g \cdot y$$

$$B_x = -g \cdot \tilde{y}$$

$$B_y = -g \cdot x$$

$$\mathbf{x} = \mathbf{x}_0 + \tilde{\mathbf{x}} \rightarrow B_y = -g \cdot \mathbf{x}_0 - g \cdot \tilde{\mathbf{x}}$$

dipole component ↑

Sources for Orbit Errors

● *Quadrupole offset:*

■ *alignment* *+/- 0.1 mm*

■ *ground motion*

■ *slow drift*

■ *civilisation*

■ *moon*

■ *seasons*

■ *civil engineering*

● *Error in dipole strength*

■ *power supplies*

■ *calibration*

● *Energy error of particles*

■ *injection energy (RF off)*

■ *RF frequency*

■ *momentum distribution*

Closed Orbit Response

■ **inhomogeneous equation:**

$$\frac{d^2 \mathbf{x}}{d s^2} + \mathbf{K}(s) \cdot \mathbf{x} = \mathbf{G}(s); \quad \mathbf{G}(s) = \frac{\mathbf{F}(s)_{\text{Lorentz}}}{\mathbf{v} \cdot \mathbf{p}_0}$$

$$\longrightarrow \underline{y}' + \begin{pmatrix} 0 & 1 \\ \mathbf{K} & 0 \end{pmatrix} \cdot \underline{y} = \underline{\mathbf{G}}; \quad \underline{\mathbf{G}} = \begin{pmatrix} 0 \\ \mathbf{G} \end{pmatrix}$$

$$\longrightarrow \underline{y}(s) = a \cdot \underline{\mathbf{S}}(s) + b \cdot \underline{\mathbf{C}}(s) + \underline{\psi}(s)$$

we need to find only one solution!

■ **variation of the constant:**

$$\underline{\psi}(s) = \underline{\varphi}(s) \cdot u(s); \quad \underline{\varphi}(s) = c \cdot \underline{\mathbf{S}}(s) + d \cdot \underline{\mathbf{C}}(s);$$

Closed Orbit Response

variation of the constant in matrix form:

$$\underline{\psi}(s) = \underline{\varphi}(s) \cdot \underline{u}(s); \quad \text{with}$$

$$\underline{\varphi}(s) = \begin{pmatrix} \sqrt{\beta(s)} \cdot \sin(\phi(s) + \phi_0) & \sqrt{\beta(s)} \cdot \cos(\phi(s) + \phi_0) \\ \cos(\phi(s) + \phi_0) / \sqrt{\beta(s)} & -\sin(\phi(s) + \phi_0) / \sqrt{\beta(s)} \end{pmatrix}$$

substitute into differential equation:

$$\longrightarrow \underline{\varphi}(s) \cdot \underline{u}'(s) = \underline{G}(s)$$

$$\longrightarrow \underline{u}(s) = \int_{s_0}^s \underline{\varphi}(t)^{-1} \cdot \underline{G}(t) dt$$

$$\longrightarrow \underline{y}(s) = \underline{a} \cdot \underline{S}(s) + \underline{b} \cdot \underline{C}(s) + \underline{\varphi}(s) \cdot \int_{s_0}^s \underline{\varphi}(t)^{-1} \cdot \underline{G}(t) dt$$

Closed Orbit Response

periodic boundary conditions:

$$\underline{y}(s) = a \cdot \underline{S}(s) + b \cdot \underline{C}(s) + \underline{\varphi}(s) \cdot \int_{s_0}^s \underline{\varphi}(t)^{-1} \cdot \underline{G}(t) dt$$

with

$$\underline{y}(s) = \begin{pmatrix} x(s) \\ x'(s) \end{pmatrix}; \quad x(s) = x(s + L); \quad x'(s) = x'(s + L)$$



periodic boundary conditions determine coefficients a and b

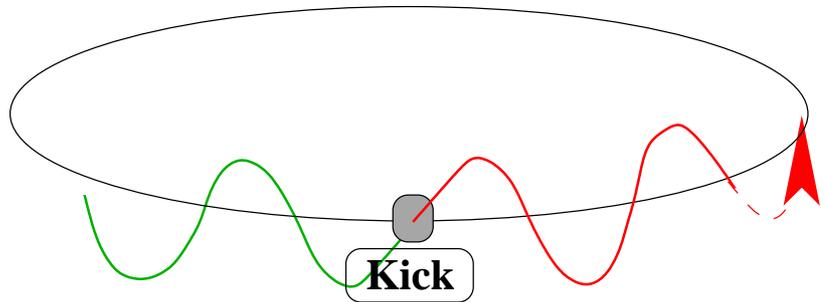


$$x(s) = \frac{\sqrt{\beta(s)}}{2 \sin(\pi \cdot Q)} \cdot \int_{s_0}^{s_0 + \text{circ}} \sqrt{\beta(t)} \cdot G(t) \cos[\phi(t) - \phi(s) - \pi Q] dt$$

Dipole Error and Orbit Stability

● Q: number of β -oscillations per turn

■ Q = N

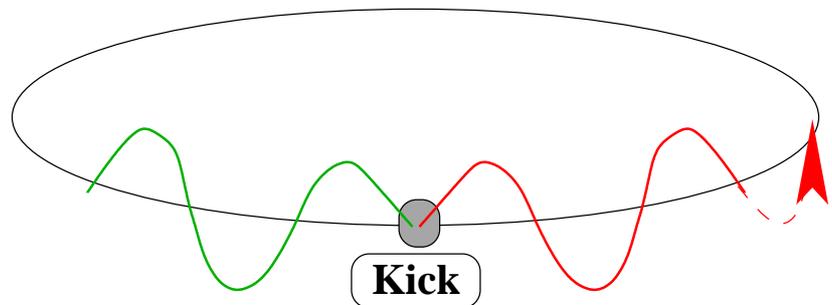


the perturbation adds up



watch out for integer tunes!

■ Q = N + 0.5



*the perturbation cancels
after each turn*

$$x(s) = \frac{\sqrt{\beta(s)}}{2 \sin(\pi \cdot Q)} \cdot \oint \sqrt{\beta(t)} \cdot G(t) \cos[|\phi(t) - \phi(s)| - \pi Q] dt$$

Closed Orbit Response

■ **Example:**

$$x(s) = \frac{-\sqrt{\beta(s)}}{2 \sin(\pi \cdot Q)} \cdot \oint \frac{-\sqrt{\beta(t)} \cdot G(t) \cos[|\phi(t) - \phi(s)| - \pi Q]}{\rho(t)} dt$$

with

$$G(t) = \frac{-1}{\rho(t)} \cdot \frac{\Delta p}{p}$$



$$x(s) = D(s) \cdot \frac{\Delta p}{p}$$

with

$$D(s) = \frac{-\sqrt{\beta(s)}}{2 \sin(\pi \cdot Q)} \cdot \oint \frac{-\sqrt{\beta(t)}}{\rho(t)} \cdot \cos[|\phi(t) - \phi(s)| - \pi Q] dt$$



Dispersion Orbit

β – Beat

■ **quadrupole error:**

$$\Delta k(s) = 0.3 \cdot \frac{\Delta B [T/m]}{p_0 [GeV]}$$

$$\frac{d^2 \mathbf{x}}{ds^2} + \mathbf{K}(s) \cdot \mathbf{x} = \mathbf{x} \cdot \Delta k(s);$$

→ **variation of the constant:**

$$\Delta \beta(s) = \frac{\beta(s)}{2 \sin(2\pi \cdot Q)} \cdot \int_{s_0}^{s_0 + \text{circ}} \beta(t) \cdot \Delta k(t) \cos[2[\phi(t) - \phi(s)] - 2\pi Q] dt$$

→

β – beat oscillates with twice the betatron frequency

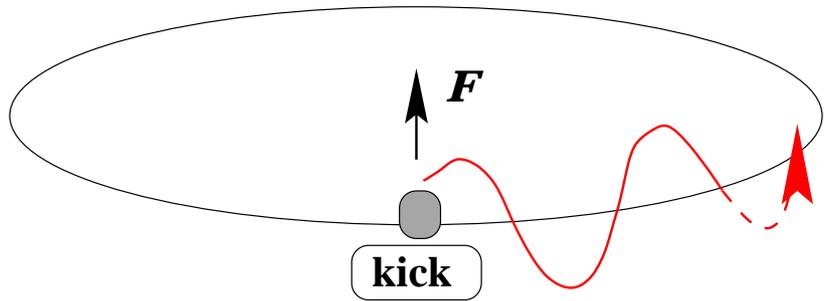
Quadrupole Error and Orbit Stability

● Quadrupole Error:

→ orbit kick proportional to
beam offset in quadrupole

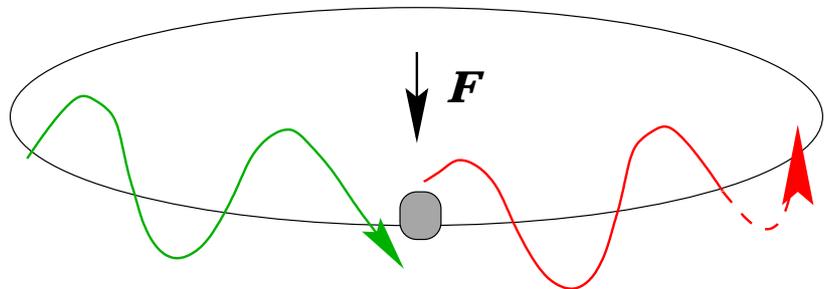
■ $Q = N + 0.5$

1. Turn: $x > 0$



→ amplitude increase

2. Turn: $x < 0$



→ amplitude increase

↘ watch out for half integer tunes!

Tune Error

one turn map:

cosine- and sine-like solutions to Hill's equation

$$\longrightarrow \underline{z}_{n+1} = M \cdot \underline{z}_n \quad \underline{z} = \begin{pmatrix} x \\ x' \end{pmatrix}$$

with

$$M = I \cdot \cos(2\pi Q) + J \cdot \sin(2\pi Q)$$

$$I = \begin{pmatrix} 1 & 0 \\ 0 & 1 \end{pmatrix} \quad J = \begin{pmatrix} \alpha & \beta \\ -\gamma & -\alpha \end{pmatrix}; \quad \gamma = [1 + \alpha^2] / \beta$$

remember:

$$\cos(2\pi Q) = \frac{1}{2} \text{trace } M$$

$$\longrightarrow \text{the coefficients of: } \frac{M - I \cdot \cos(2\pi Q)}{\sin(2\pi Q)}$$

provide the optic functions at s_0

Tune Error

transfer matrix for single quadrupole:

$$m_0 = \begin{pmatrix} 1 & 0 \\ -k \cdot l & 1 \end{pmatrix}$$

matrix for single quadrupole with error:

$$m = \begin{pmatrix} 1 & 0 \\ [-k + \Delta k] \cdot l & 1 \end{pmatrix}$$

one turn matrix with quadrupole error:

$$M = m \cdot m_0^{-1} \cdot M_0$$

trace M



$$\cos(2\pi Q) = \cos(2\pi Q_0) - \frac{1}{2} \beta \cdot \Delta k \cdot l \sin(2\pi Q_0)$$

Tune Error

distributed perturbation:

$$\cos(2\pi Q) = \cos(2\pi Q_0) - \frac{\sin(2\pi Q_0)}{2} \cdot \int \beta \cdot \Delta k \, ds$$

$$\longrightarrow \Delta Q = \frac{1}{4\pi} \cdot \int \beta \cdot \Delta k \, ds$$

chromaticity:

$$k_0 = \frac{e \cdot g}{p}$$

momentum error $\longrightarrow \Delta k = -k_0 \cdot \frac{\Delta p}{p}$

$$\begin{aligned} \Delta Q &= -\frac{1}{4\pi} \cdot \int \beta \cdot k_0 \, ds \cdot \frac{\Delta p}{p} \\ &= \xi \cdot \frac{\Delta p}{p} \end{aligned}$$

Problems Generated by Orbit Errors

● injection errors:

■ *aperture* → *beam losses*

■ *filamentation* → *beam size*

● closed orbit errors:

■ *x-y coupling*

■ *aperture*

■ *energy error*

■ *field imperfections*

■ *dispersion* → *beam size at IP*

■ *beam separation*

Aim:

$\Delta x, \Delta y < 4 \text{ mm}$

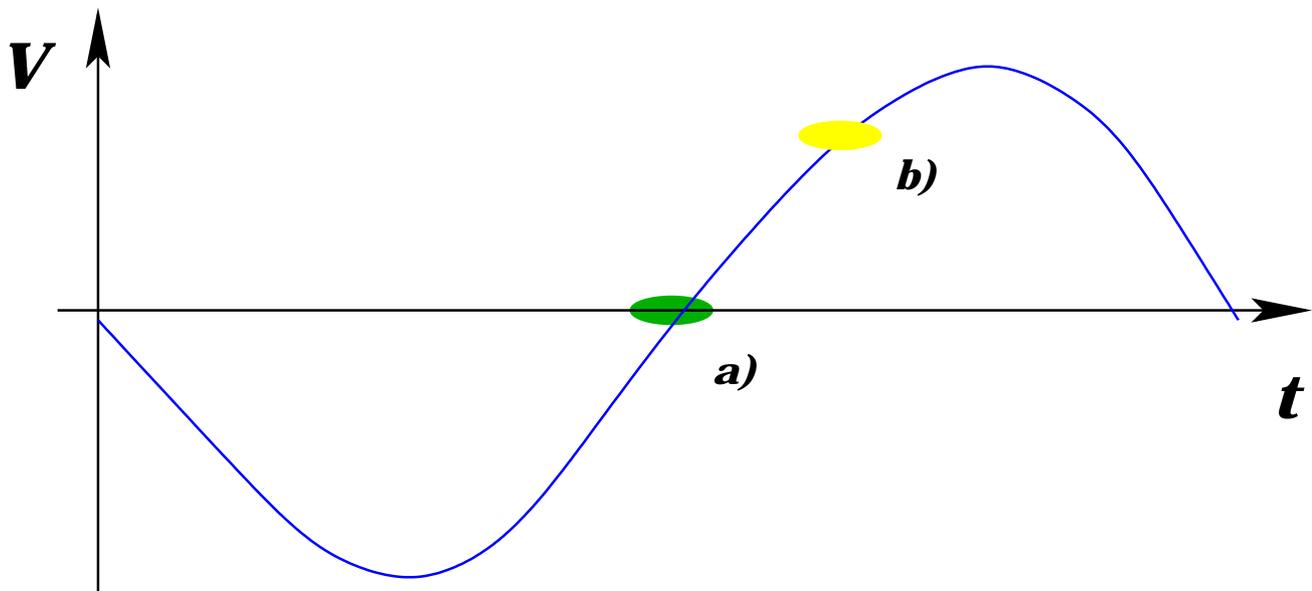
rms < 0.5 mm

→ *beam monitors and orbit correctors*

Synchrotron:

→ ***the orbit determines the particle energy!***

■ ***assume: $L >$ design orbit***



→ ***energy increase***

Equilibrium:

$$f_{RF} = h \cdot f_{rev}$$

$$f_{rev} = \frac{1}{2 \cdot \pi} \cdot \frac{q}{m \cdot \gamma} \cdot B$$

→ ***E depends on orbit and magnetic field!***

Orbit Correction

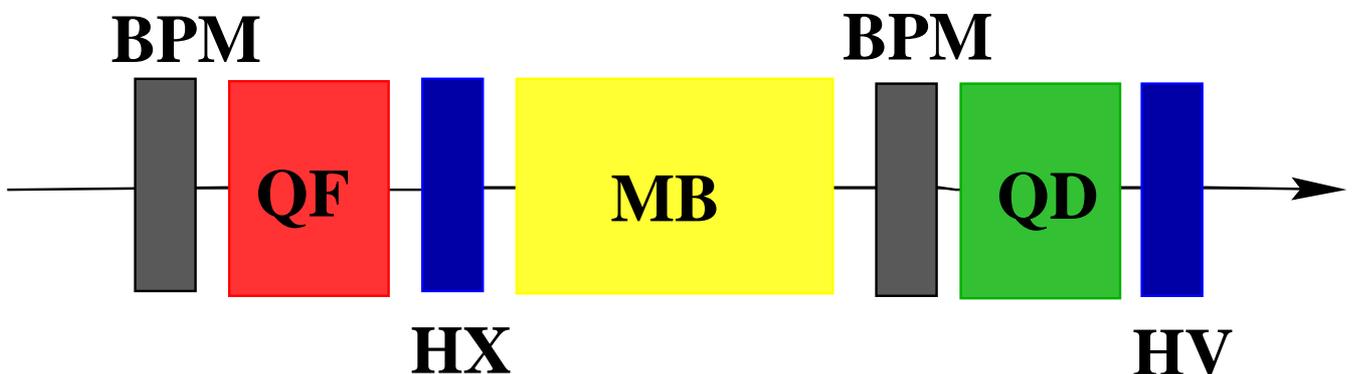
- the orbit error in a storage ring with conventional magnets is dominated by the contributions from the quadrupole alignment errors
- orbit perturbation is proportional to the local β -functions at the location of the dipole error
 - alignment errors at QF cause mainly horizontal orbit errors
 - alignment errors at QD causes mainly vertical orbit errors

Orbit Correction

■ aim at a local correction of the dipole error due to the quadrupole alignment errors

→ place orbit corrector and BPM next to the main quadrupoles

→ horizontal BPM and corrector next to QF
vertical BPM and corrector next to QD



→ orbit in the opposite plane?

relative alignment of BPM and quadrupole?

Local Orbit Bumps I

deflection angle:

$$\theta_i = \int_{\text{dipole}} G_i(t) dt = \frac{0.3 \cdot B_i[\text{T}] \cdot l}{p[\text{GeV}]}$$

trajectory response:

[no periodic boundary conditions]

$$\longrightarrow x(s) = \sqrt{\beta_i \beta(s)} \cdot \theta_i \cdot \sin[\phi(s) - \phi_i]$$

$$\longrightarrow x'(s) = \sqrt{\beta_i / \beta(s)} \cdot \theta_i \cdot \cos[\phi(s) - \phi_i]$$

Local Orbit Bumps II

closed orbit bump:

compensate the trajectory perturbation with

additional corrector kicks further down stream

→ closure of the perturbation within one turn

→ local orbit excursion

→ possibility to correct orbit errors locally

→ closure with one additional corrector magnet

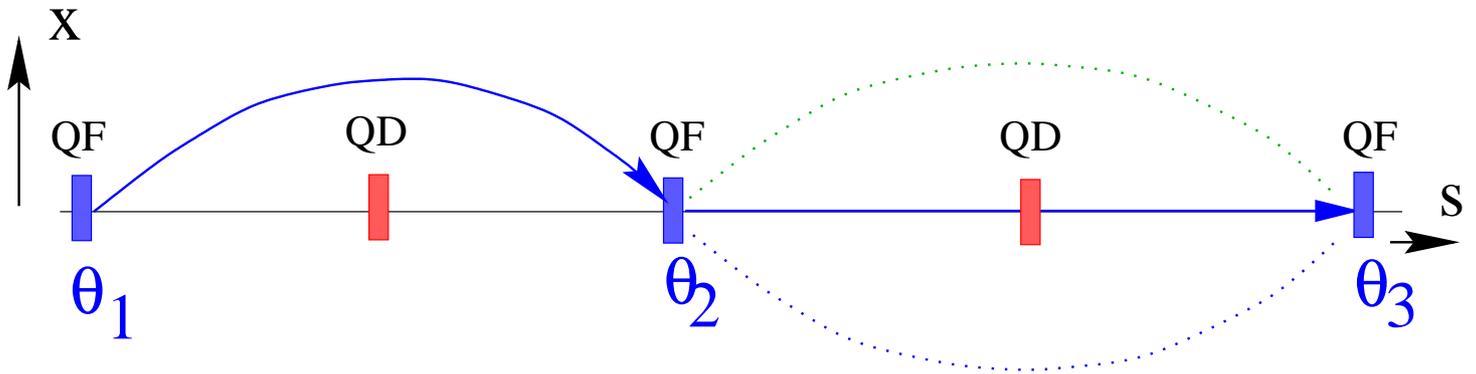
→ π - bump

→ closure with two additional corrector magnets

→ three corrector bump

Local Orbit Bumps III

■ π - bump: (quasi local correction of error)



→
$$\theta_2 = \frac{\sqrt{\beta_1}}{\sqrt{\beta_2}} \cdot \theta_1$$

■ limits / problems:

→ closure depends on lattice phase advance

→ requires 90° lattice

→ sensitive to lattice errors

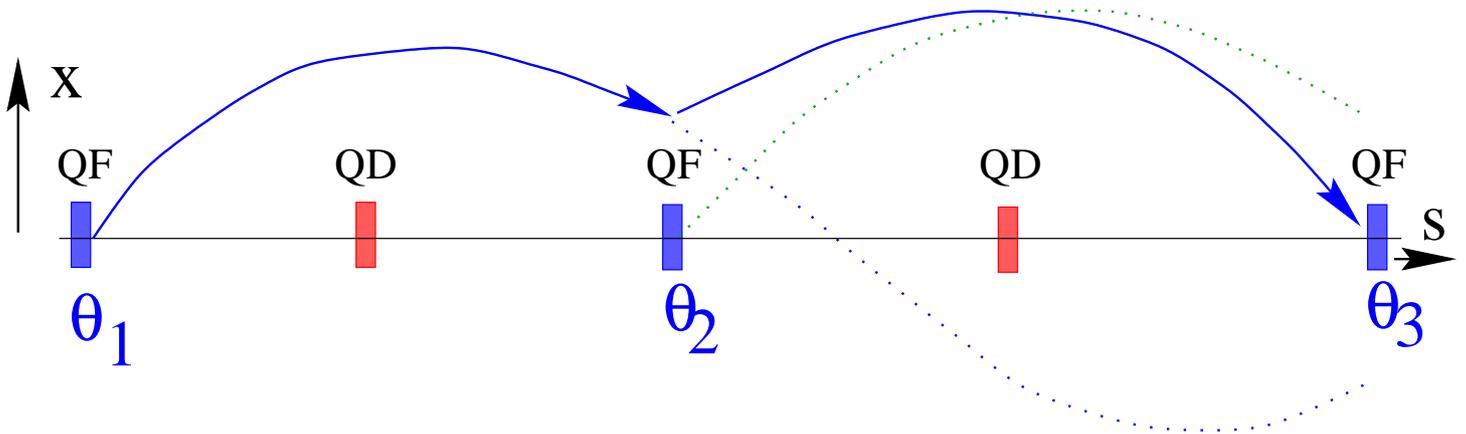
→ requires horizontal BPMs at QF and QD

→ sensitive to BPM errors

→ requires large number of correctors

Local Orbit Bumps IV

3 corrector bump: (quasi local correction of error)



$$\theta_2 = -\frac{\sqrt{\beta_1}}{\sqrt{\beta_2}} \cdot \frac{\sin(\Delta\phi_{3-1})}{\sin(\Delta\phi_{3-2})} \cdot \theta_1$$

$$\theta_3 = \left(\frac{\sin(\Delta\phi_{3-1})}{\tan(\Delta\phi_{3-2})} - \cos(\Delta\phi_{3-1}) \right) \cdot \frac{\sqrt{\beta_1}}{\sqrt{\beta_3}} \cdot \theta_1$$

works for any lattice phase advance

requires only horizontal BPMs at QF

limits / problems:

sensitive to BPM errors

large number of correctors

can not control x'

Harmonic Filtering I

unperturbed solution (smooth approximation):

$$x'' + \left[\frac{2\pi}{C} \cdot Q\right]^2 x = 0 \quad \longrightarrow \quad x(s) = A \cdot e^{i \frac{2\pi}{C} \cdot Q \cdot s}$$

orbit perturbation due to random kicks:

$$x'' + K(s) \cdot x = \sum_{i=1}^m \theta_i \cdot \delta(s-s_i) \\ = F(s)$$

periodic boundary conditions:

$$\longrightarrow \quad x(s) = \sum_n d_n \cdot e^{i \frac{2\pi}{C} \cdot n \cdot s}$$

$$\longrightarrow \quad F(s) = \sum_n f_n \cdot e^{i \frac{2\pi}{C} \cdot n \cdot s}$$

Harmonic Filtering II

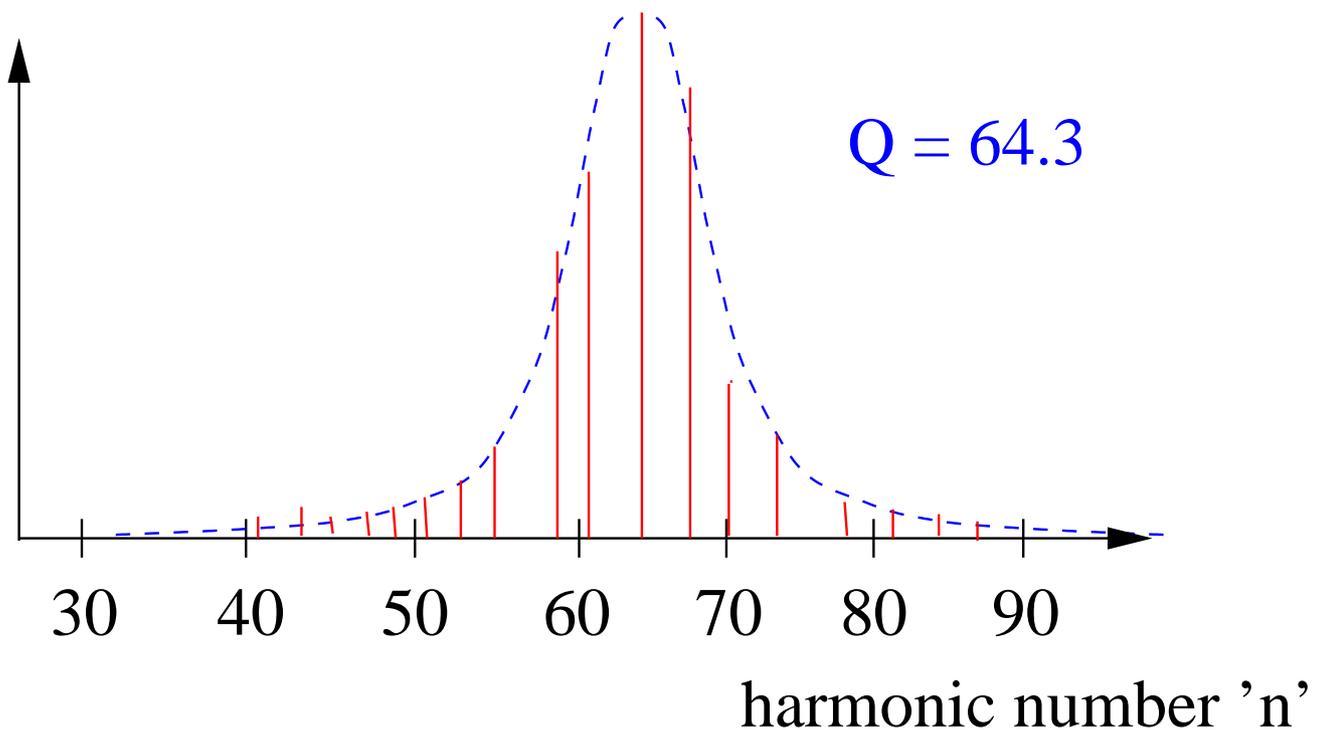
inserting Ansatz into Hill's equation:

$$\rightarrow x(s) = \sum_n d_n \cdot e^{i \frac{2\pi}{C} \cdot n \cdot s}$$

with:

$$d_n = \frac{f_n}{\left(\frac{2\pi}{C} \cdot Q\right)^2 - \left(\frac{2\pi}{C} \cdot n\right)^2}$$

→ the spectrum peaks around the tune



→ small number of correctors are efficient

SVD I

linear relation between BPM and corrector data:

COR: vector of corrector amplitudes

$$\longrightarrow \underline{\text{COR}} = [a_1, a_2, \dots, a_m]$$

BPM: vector of all BPM data

$$\longrightarrow \underline{\text{BPM}} = [b_1, b_2, \dots, b_n]$$

$$\longrightarrow \underline{\text{BPM}} = A \cdot \underline{\text{COR}}; \quad A = n \times m \text{ matrix}$$

global orbit correction:

ORB: vector of all measured orbit data

$$\longrightarrow \underline{\text{ORB}} = [c_1, c_2, \dots, c_n]$$

find a set of corrector settings that satisfies:

$$\underline{\text{ORB}} - A \cdot \underline{\text{COR}} = \underline{0}$$

SVD II

mathematical solution:

$$\underline{\text{COR}} = \underline{A}^{-1} \cdot \underline{\text{ORB}}$$

problem:

A is normally not invertible

(A is normally not even a square matrix)

→ minimise the norm: $\| \underline{\text{ORB}} - \underline{A} \cdot \underline{\text{COR}} \|$

$$\text{with } \| \underline{\mathbf{X}} \| = \left(\sum_{i=1}^m |x_i|^p \right)^{1/p}$$

→ find a matrix B such that:

$$\| \underline{\text{ORB}} - \underline{A} \cdot \underline{B} \cdot \underline{\text{ORB}} \|$$

attains a minimum

SVD III

■ singular value decomposition SVD:

$$\longrightarrow A = O_1 \cdot D \cdot O_2$$

O_1 and O_2 are orthogonal matrices: $O^{-1} = O^t$

$$(a_{ij} \longrightarrow a_{ji})$$

D is a diagonal matrix:

$$D = \begin{pmatrix} \sigma_{11} & & 0 & \dots & 0 \\ & \sigma_{22} & & & \\ 0 & & \dots & & \\ & & & \sigma_{kk} & \\ & & & & 0 \end{pmatrix}$$

$$k \leq \min(n, m)$$

define:

$$\hat{D} = \begin{pmatrix} 1/\sigma_{11} & & 0 \\ & 1/\sigma_{22} & \\ 0 & & \dots \\ & & & 1/\sigma_{kk} \\ & & & & 0 \end{pmatrix}$$

$$\longrightarrow D \cdot \hat{D} = \begin{pmatrix} 1 & & 0 \\ & \dots & \\ 0 & & & 1 \end{pmatrix}$$

SVD IV

define the correction matrix:

$$\rightarrow B = O_2^t \cdot \hat{D} \cdot O_1^t$$

$$\rightarrow A \cdot B = (O_1 \cdot D \cdot O_2) \cdot (O_2^t \cdot \hat{D} \cdot O_1^t)$$

SVD allows you to adjust k corrector magnets

$$k = \min(m, n)$$

- if $k = m = n$ one obtains a zero orbit
(by using all possible corrector magnets)
- if $m \neq n$ SVD algorithm minimises the norm
(by using all possible corrector magnets)
- algorithm is not stable if \underline{D} has small eigenvalues

Most Effective Corrector

■ orbit is perturbed by a few large perturbations:

→ minimise

$$\| \text{BPM} - \text{A} \cdot \text{COR} \| \quad \text{with} \quad \| \underline{\mathbf{X}} \| = \left(\sum_{i=1}^m |x_i|^p \right)^{1/p}$$

with a small set of 'k' corrector magnets

■ brut force: select all possible combinations

→ time consuming but good result

■ selective: keep the already selected correctors

→ much faster!

→ finite chance to miss best choice

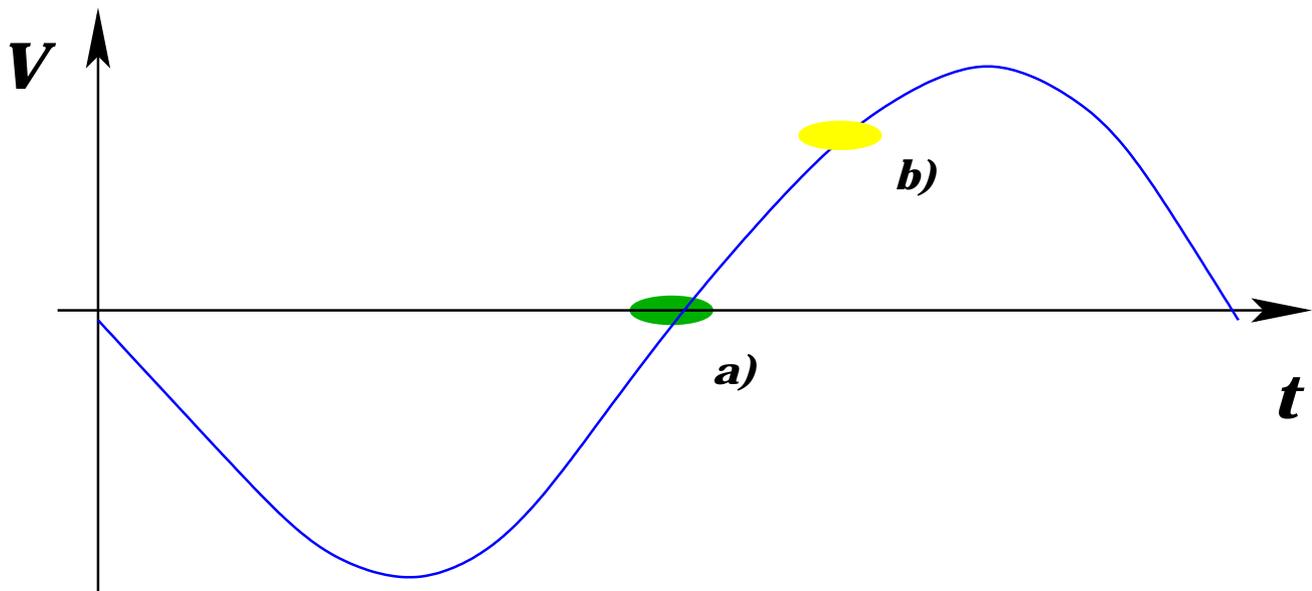
→ can generate orbit bumps

■ MICADO: selective + cross-correlation between orbit residues and remaining correctors

Synchrotron:

→ ***the orbit determines the particle energy!***

■ ***assume: $L >$ design orbit***



→ ***energy increase***

Equilibrium:

$$f_{RF} = h \cdot f_{rev}$$

$$f_{rev} = \frac{1}{2 \cdot \pi} \cdot \frac{q}{m \cdot \gamma} \cdot B$$

→ ***E depends on orbit and magnetic field!***